

OpenFOAM Analysis on the Thermal-Hydraulics in the CANDU-6 Moderator



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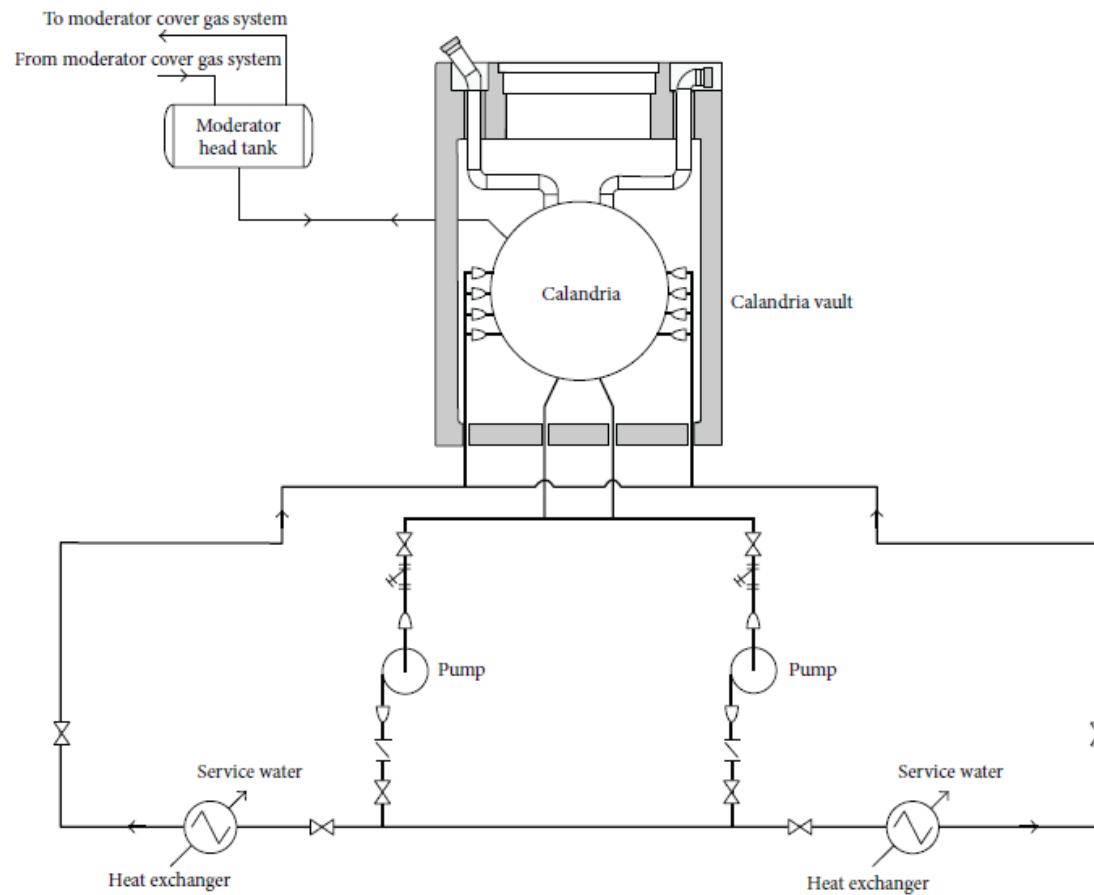
** Korea Atomic Energy Research Institutue

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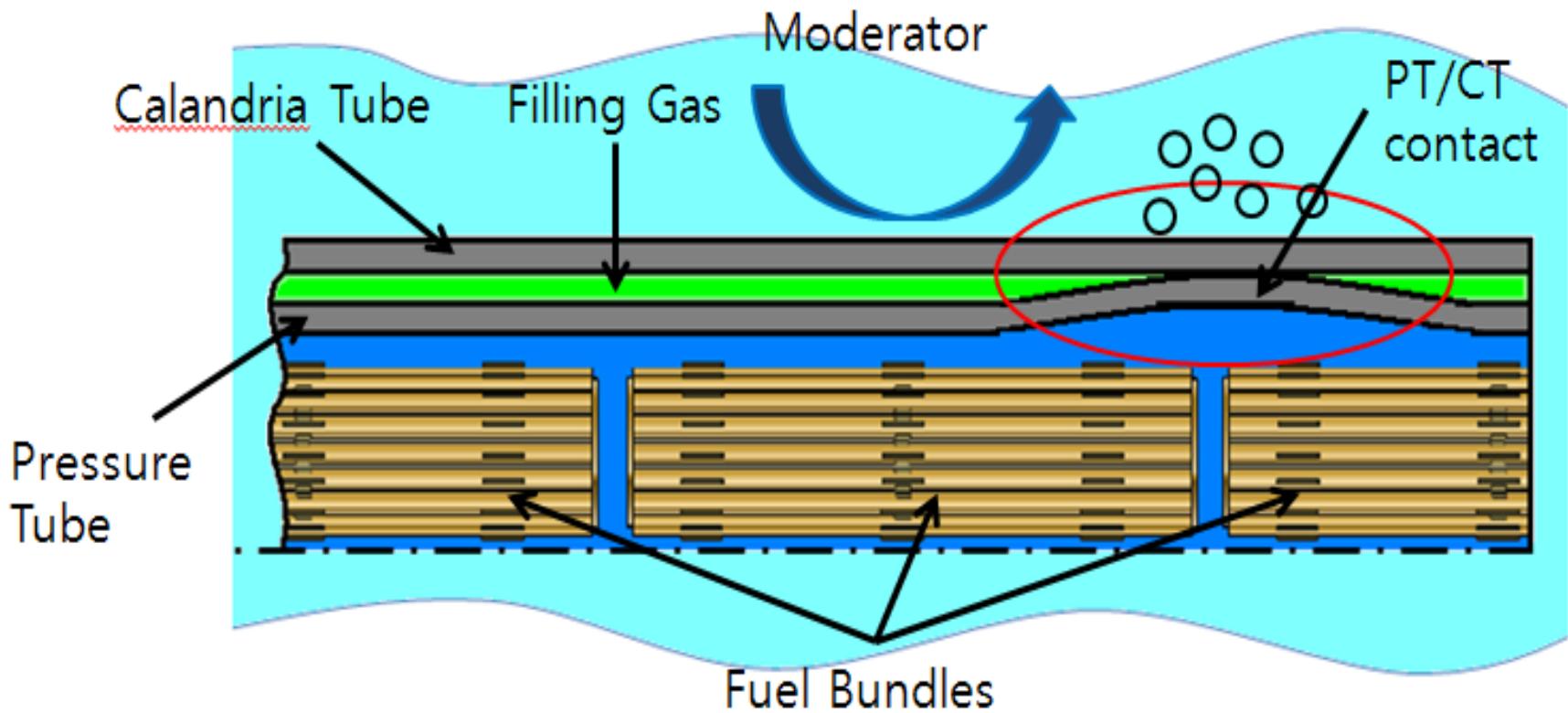
- Objective of Research
- Benchmark: STERN Lab. Experiment
- CANDU-6 Moderator Circulation
- Summary

Objective of Research

3-D Analysis with CFD(Computational Fluid Dynamics code and consideration of its feasibility in the moderator system of heavy-water CANDU reactors: Calandria tank



Introduction: Calandria Tubes



LOCA: Loss Of Coolant Accident

CHF: Critical Heat Flux

Codes: MODTRUC/MODTRUC_CLAS, CATHENA, CUPID

Key Physics

Primary Physical Phenomena

- 1) Incompressible flow solver + turbulence model
- 2) Heat transfer: energy equation
- 3) Buoyancy term: source terms in N-S equation
- 4) Porous media: modeling of porosity, permeability
- 5) Heat source model: energy equation

Boundary conditions

No-slip (wall), adiabatic, inlet/outlet,

Governing Equations

$$\nabla \cdot \mathbf{V} = 0$$

$$\rho \left\{ \frac{\partial \mathbf{V}}{\partial t} + (\mathbf{V} \cdot \nabla) \mathbf{V} \right\} = -\nabla p + \rho \mathbf{g} + \mu \nabla^2 \mathbf{V} + \boxed{\rho_0 \mathbf{g} \beta (T_0 - T)}$$

$$\rho C_p \left\{ \frac{\partial T}{\partial t} + (\mathbf{V} \cdot \nabla) T \right\} = \sigma \nabla^2 T + \boxed{Q_s}$$

$$\rho \left\{ \frac{\partial k}{\partial t} + (\mathbf{V} \cdot \nabla) k \right\} = \frac{\partial}{\partial x_j} \left\{ \left(\mu + \frac{\mu_t}{\sigma_k} \right) \frac{\partial k}{\partial x_j} \right\} + P_k + P_b - \rho \varepsilon - Y_M + S_k$$

$$\rho \left\{ \frac{\partial \varepsilon}{\partial t} + (\mathbf{V} \cdot \nabla) \varepsilon \right\} = \frac{\partial}{\partial x_j} \left\{ \left(\mu + \frac{\mu_t}{\sigma_\varepsilon} \right) \frac{\partial \varepsilon}{\partial x_j} \right\} + C_{1\varepsilon} \frac{\varepsilon}{k} (P_k + C_{3\varepsilon} P_b) - C_{2\varepsilon} \rho \frac{\varepsilon^2}{k} + S_\varepsilon$$

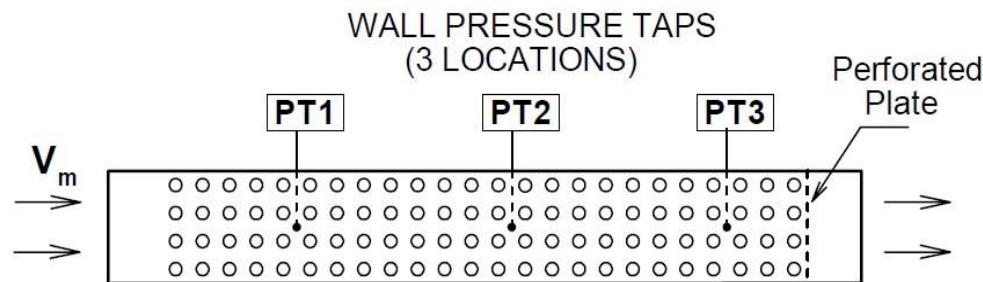
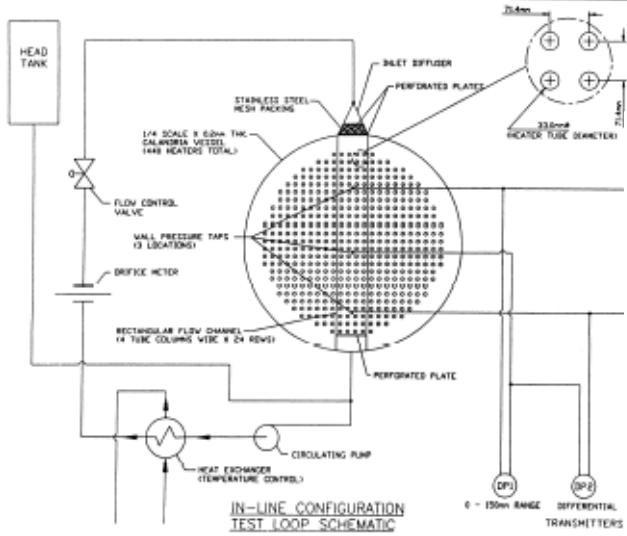
$$\mu_t = \rho C_\mu \frac{k^2}{\varepsilon} \quad C_{1\varepsilon} = 1.44, \quad C_{2\varepsilon} = 1.92, \quad C_\mu = 0.09, \quad \sigma_k = 1.0, \quad \sigma_\varepsilon = 1.3$$

Solvers

Table 1. OpenFOAM solvers used in the present computation.

Types	Name of solver	Related physics
Incompressible (isothermal)	(porous)SimpleFoam simpleFoam/pimpleFoam	(Prorous media) Steady/Unsteady-state Incompressible Turbulent (model)
Heat Transfer	buoyantBoussinesqSimpleFoam/ buoyantBoussinesqPimpleFoam	Steady/Unsteady-state Buoyant: free convection Incompressible Turbulent (model)

Benchmark: STERN Lab. Exp.



(a) In-Line Configuration

Table I Comparison of Measured and Predicted Pressure Drops

	STERN T.P. 326	STERN T.P. 299	STERN T.P. 306
Mass Flow [kg/s]	3.089	3.904	5.734
Temp. [°C]	39.5	63.6	79.8
V _m [m/s]	0.054	0.070	0.103
Reynolds Number	2746	5237	9392
ΔP [Pa]	Measurement, A	28.2	41.3
	CFX-4, B	27.6	41.2
	MODTURC	30.5	44.9
	$ A - B / A$	2.13 %	0.242 %
			0.762 %

Material Properties

Case	Vm [m/s]	Density [kg/m ³]	Viscosity [kg/(m s)]	Re _d
1	0.054	992.25	0.000653	2,709
2	0.070	981.00	0.000440	5,153
3	0.103	971.60	0.000355	9,308

Dia. of tube: 0.03302 m

X = 2 m

Y = 0.02856 m

Z = 0.2 m

Boundary Condition / Grids

Inlet Boundary Condition: fix volume flow rate

Turbulence Intensity: assume 5%

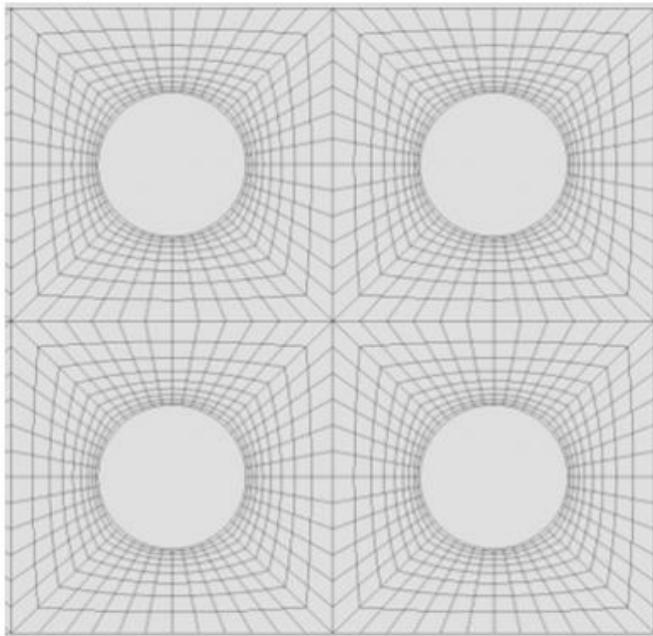
$$k = \frac{1}{2} \|\mathbf{V}\|^2$$

$$\varepsilon = \frac{C_\mu^{0.75} k^{1.5}}{l}$$

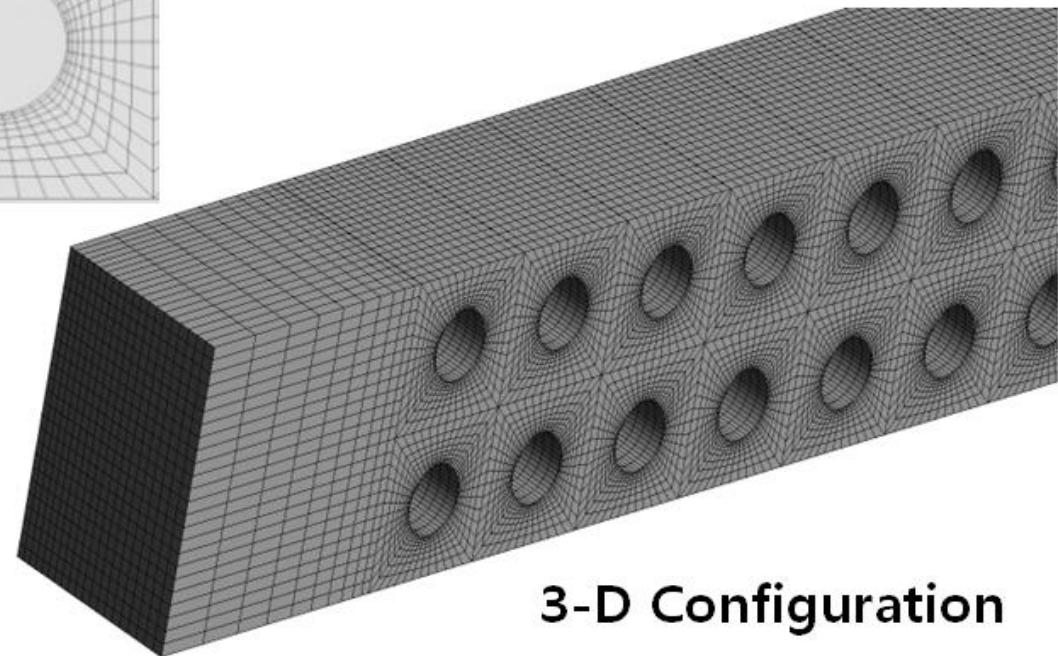
Outlet Boundary Condition: const pressure

Wall Boundary Condition: no slip

3-D Grids System

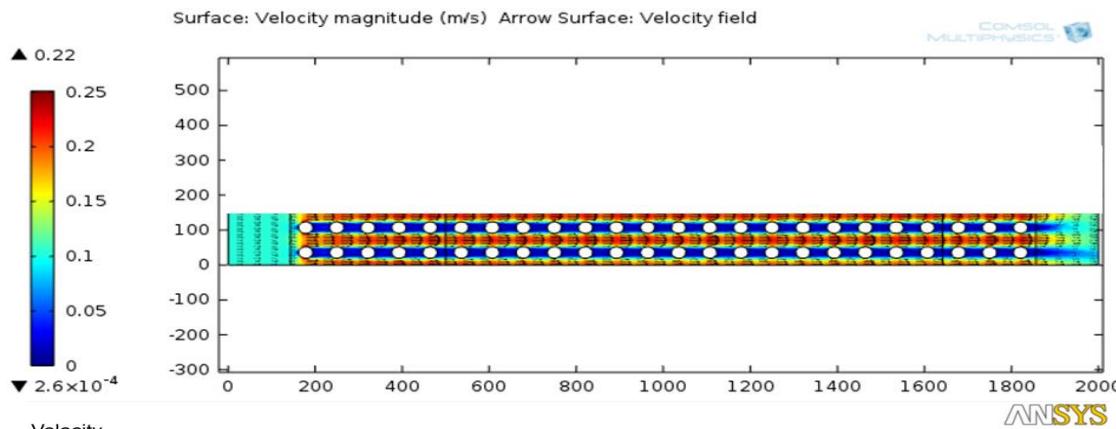


Tube Channel

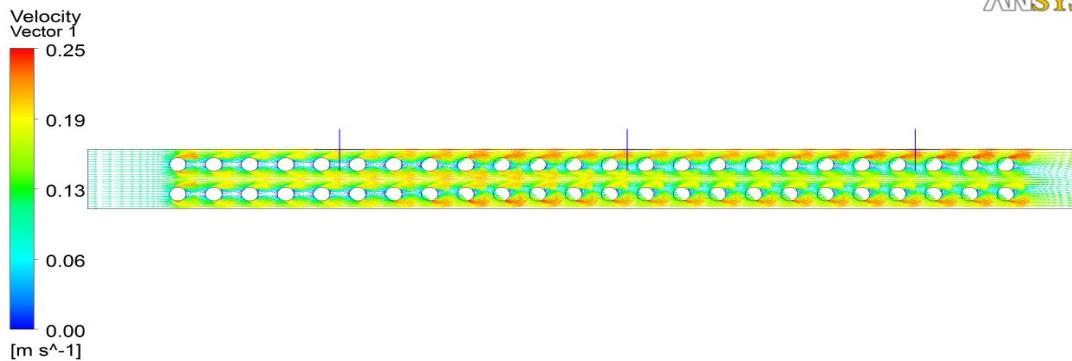


3-D Configuration

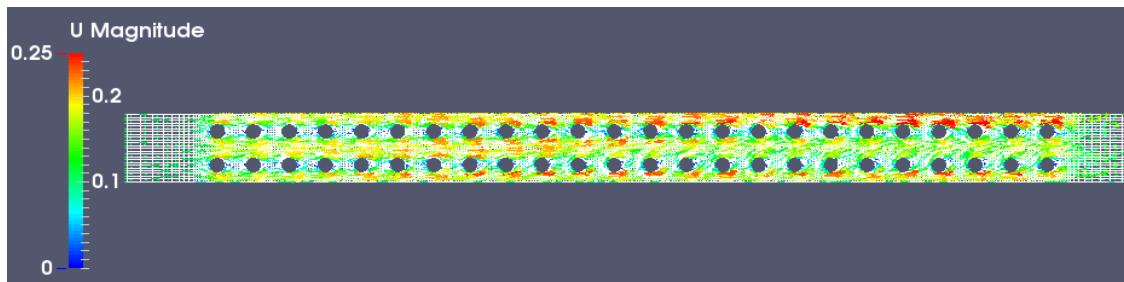
Comparison of Result



COMSOL Multiphysics

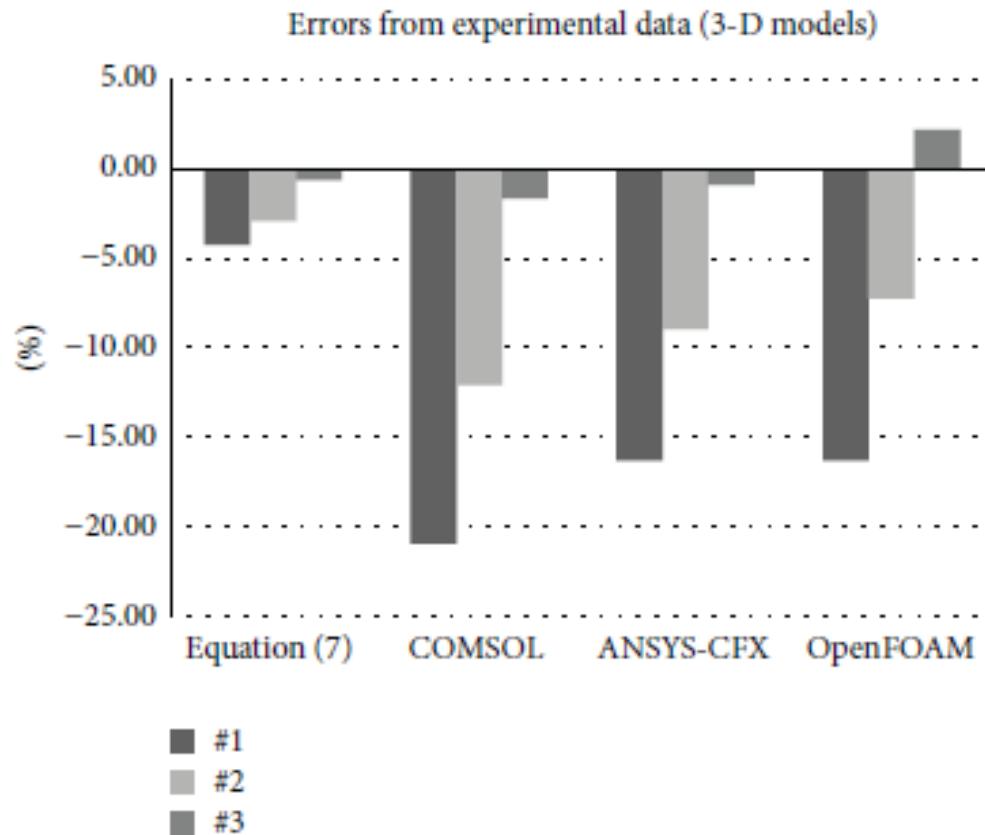


ANSYS-CFX



OpenFOAM

Comparison of Results (I)



Cause of error:

- 1) Turbulence Intensity:
 - Intake Turbulence
 - Surface Roughness
- 2) 3-D Effect:
 - Axial Flow along the Cylinder Surface
 - Relaxation of Flow Instability

Comparison of Results (II)

2-D

Case	Experiment (STERN)	COMSOL (2D)	Δp (Pa)	ANSYS-FLUENT (2D)*	OpenFOAM (2D)
#1	28.2	15.0		16.0	14.0
#2	41.3	22.5		20.3	21.4
#3	78.2	45.2		36.4	38.0

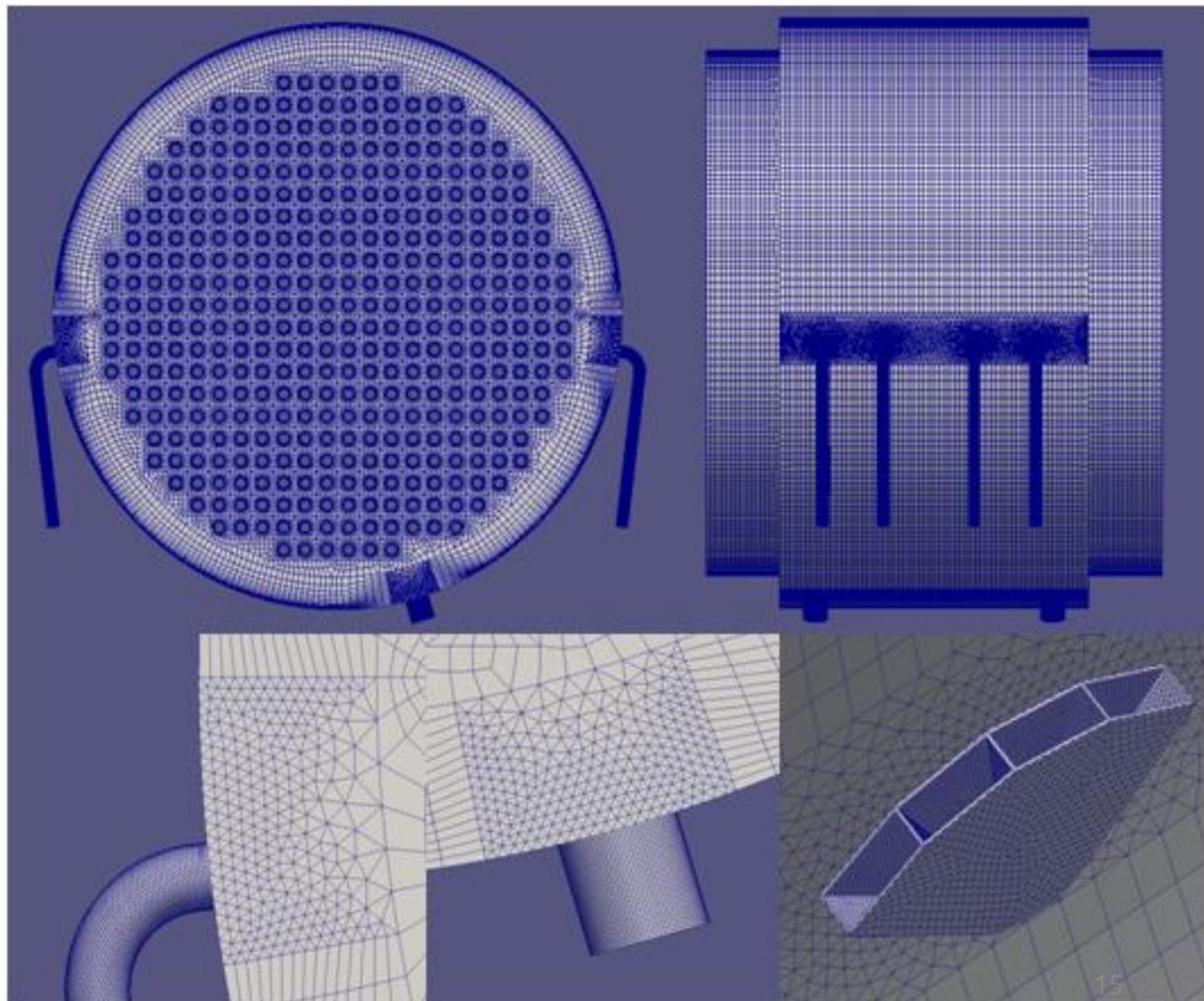
* Quoted for the data with $k-\epsilon$ turbulence model in FLUENT version 15 [10].

3-D

Case	Experiment (STERN)	Equation (7)*	Δp (Pa)	COMSOL (3-D)	ANSYS-CFX (3-D)	OpenFOAM (3-D)
#1	28.2	27.0		22.3	23.6	23.6
#2	41.3	40.1		36.3	37.6	38.3
#3	78.2	77.7		76.9	77.5	79.9

* The porous model is used with the experimental correlation [15].

Full 3-D CANDU-6 Grids



Total = 6.74 M
Hexa = 5.11 M
Pyramid = 0.013 M
Tetra = 1.62 M

Boundary/Initial Conditions

- Massflow rate at each inlet = 127.4 kg/s.
(Total Massflow = 1,019 kg/s)
- No-slip at the wall.
- Outlet pressure: fixed, & zero-gradient otherwise.
- Inlet temperature = 47.3 degC.
- Adiabatic at the wall.
- Initially stationary & isothermal.

Modeling of the Heat Source

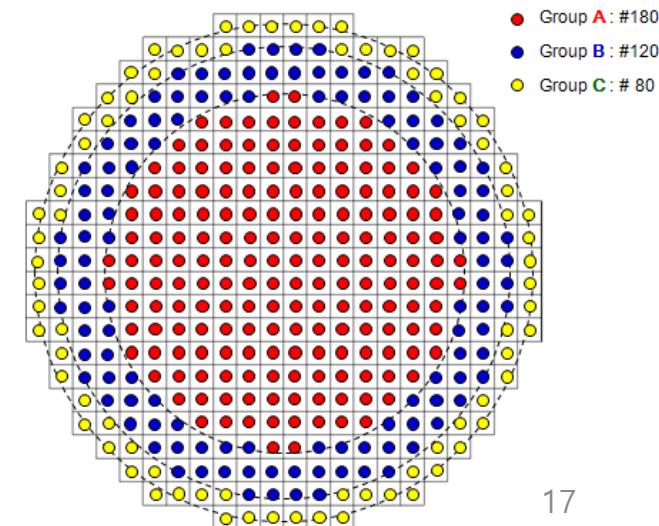
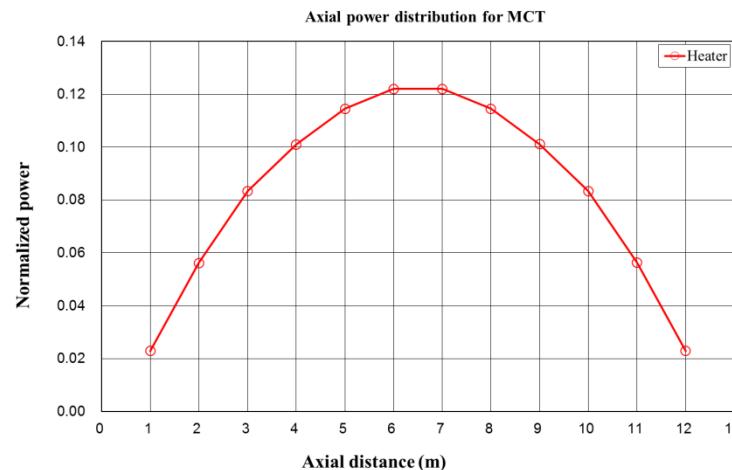
Total Thermal Power = 100 MW

Correction Factor for the Volume of Tube Bundle
= 1.089

$$Q_s(r, z) = Q_s f_r(r) f_z(z)$$

$$f_r(r) = 0.94588 - 0.01989 r + 0.0995 r^2 - 0.03888 r^3 - 0.00256 r^4 \quad (0.0 \leq r \leq 3.38 \text{ m})$$

$$f_z(z) = 1.0 - 0.1111 z^2 \quad (-3.0 \leq z \leq 3.0 \text{ m})$$



Implementation with OpenFOAM

File Name: fvOption

```
energySource1
{
    type scalarCodedSource;
    active true;
    selectionMode all;
    #{
        ...
        forAll(cc,cell)
        {
            ...
        } //define the source terms
    }
    #};
}
```

Turbulence Model: Realizable k- ϵ

$$\frac{\partial}{\partial t}(\rho k) + \frac{\partial}{\partial x_j}(\rho k u_j) = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_k} \right) \frac{\partial k}{\partial x_j} \right] + P_k + P_b - \rho \epsilon - Y_M + S_k$$

$$\frac{\partial}{\partial t}(\rho \epsilon) + \frac{\partial}{\partial x_j}(\rho \epsilon u_j) = \frac{\partial}{\partial x_j} \left[\left(\mu + \frac{\mu_t}{\sigma_\epsilon} \right) \frac{\partial \epsilon}{\partial x_j} \right] + \rho C_1 S \epsilon - \rho C_2 \frac{\epsilon^2}{k + \sqrt{\nu \epsilon}} + C_{1\epsilon} \frac{\epsilon}{k} C_{3\epsilon} P_b + S_\epsilon$$

$$C_1 = \max \left[0.43, \frac{\eta}{\eta + 5} \right], \quad \eta = S \frac{k}{\epsilon}, \quad S = \sqrt{2S_{ij}S_{ij}}$$

$$\begin{aligned} \mu_t &= \rho C_\mu \frac{k^2}{\epsilon} & C_\mu &= \frac{1}{A_0 + A_s \frac{kU^*}{\epsilon}} \\ U^* &\equiv \sqrt{S_{ij}S_{ij} + \tilde{\Omega}_{ij}\tilde{\Omega}_{ij}} & \tilde{\Omega}_{ij} &= \underline{\Omega_{ij}} - 2\epsilon_{ijk}\omega_k \\ \underline{\Omega_{ij}} &= \overline{\Omega_{ij}} - \epsilon_{ijk}\omega_k \end{aligned}$$

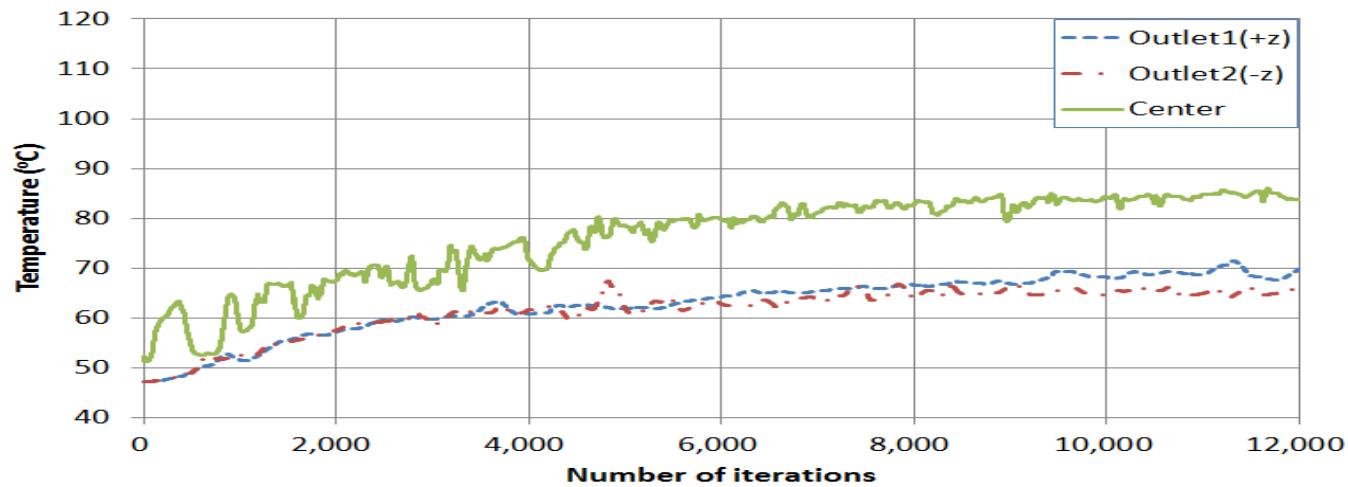
$$\begin{aligned} A_0 &= 4.04, \quad A_s = \sqrt{6} \cos \phi \\ \phi &= \frac{1}{3} \cos^{-1}(\sqrt{6}W), \quad W = \frac{S_{ij}S_{jk}S_{ki}}{\tilde{S}^3}, \quad \tilde{S} = \sqrt{S_{ij}S_{ij}}, \quad S_{ij} = \frac{1}{2} \left(\frac{\partial u_j}{\partial x_i} + \frac{\partial u_i}{\partial x_j} \right) \end{aligned}$$

$$C_{1\epsilon} = 1.44, \quad C_2 = 1.9, \quad \sigma_k = 1.0, \quad \sigma_\epsilon = 1.2$$

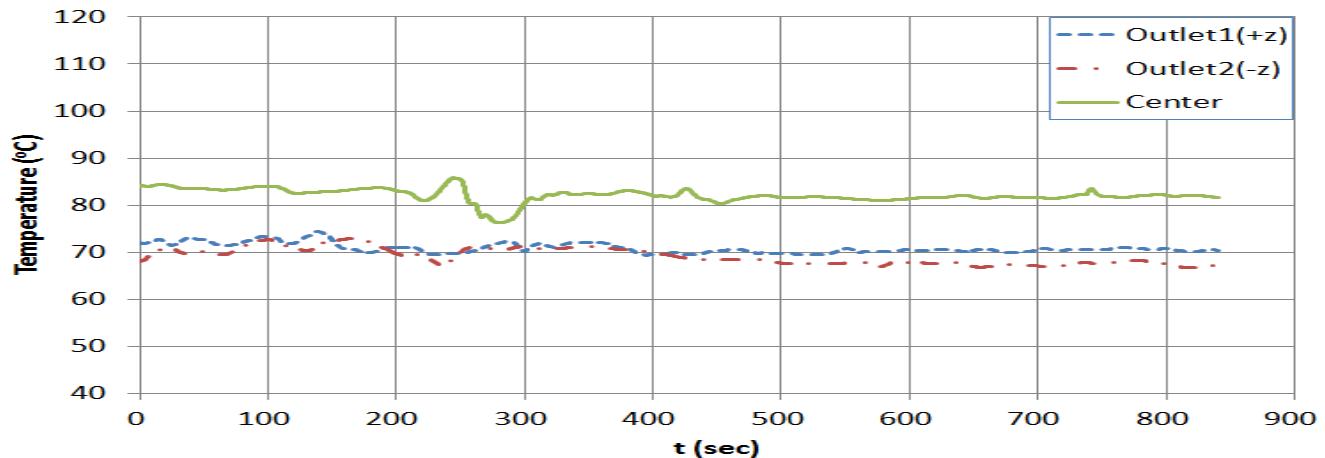
Complex rotational flow

Temperature at the Inlet/Outlet

Stage 1

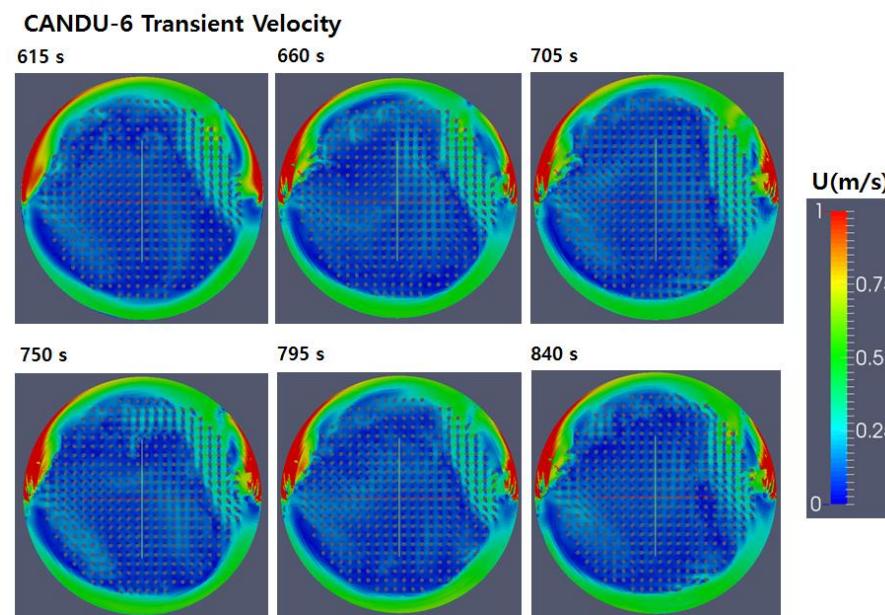


Stage 2

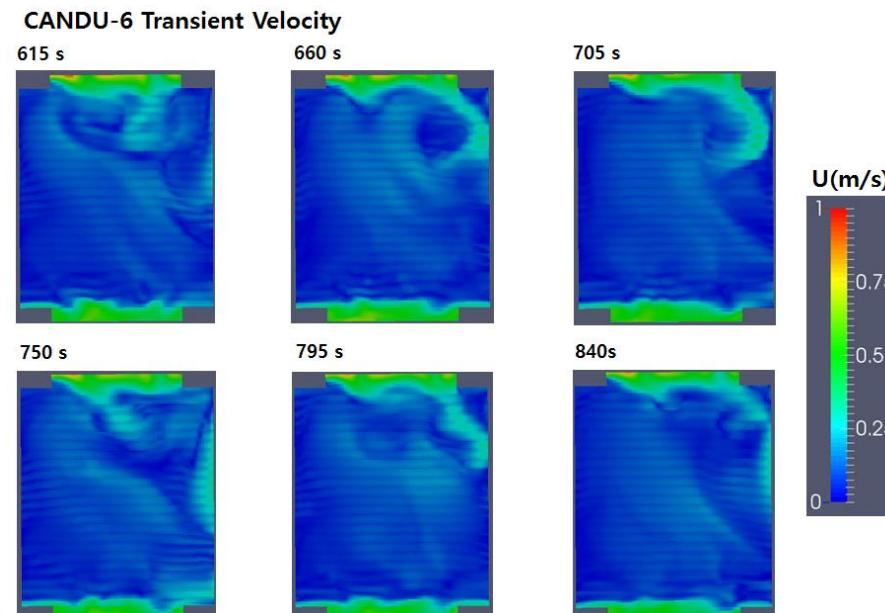


Velocity Profiles

Cross-sectional

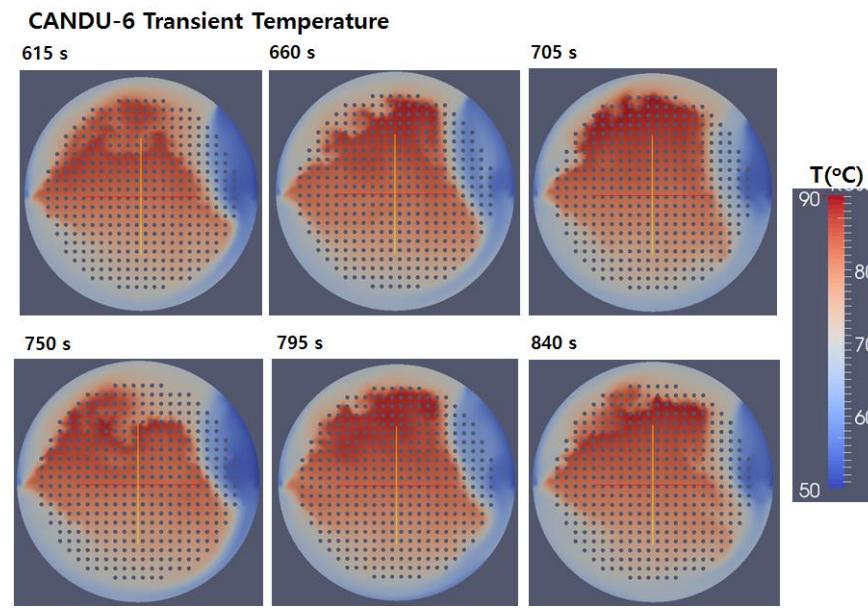


Axial

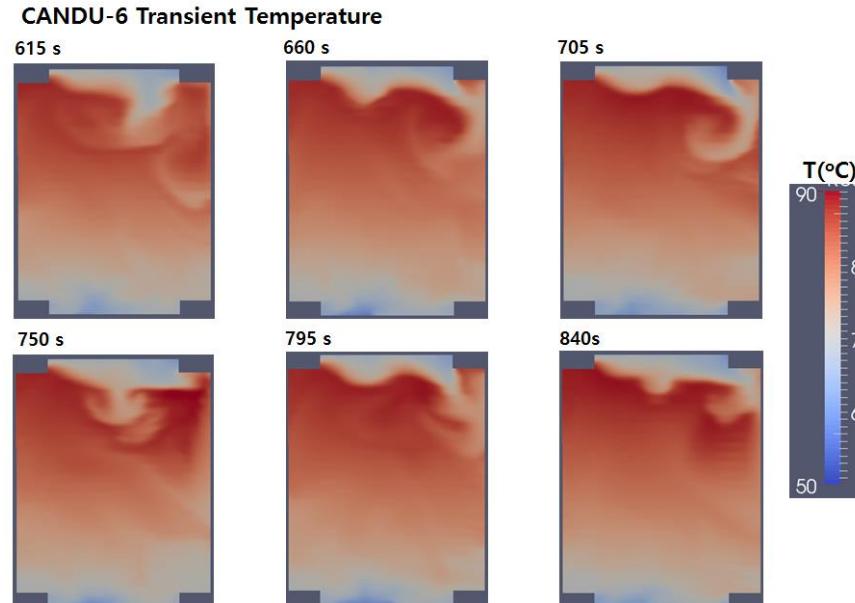


Temperature Profiles

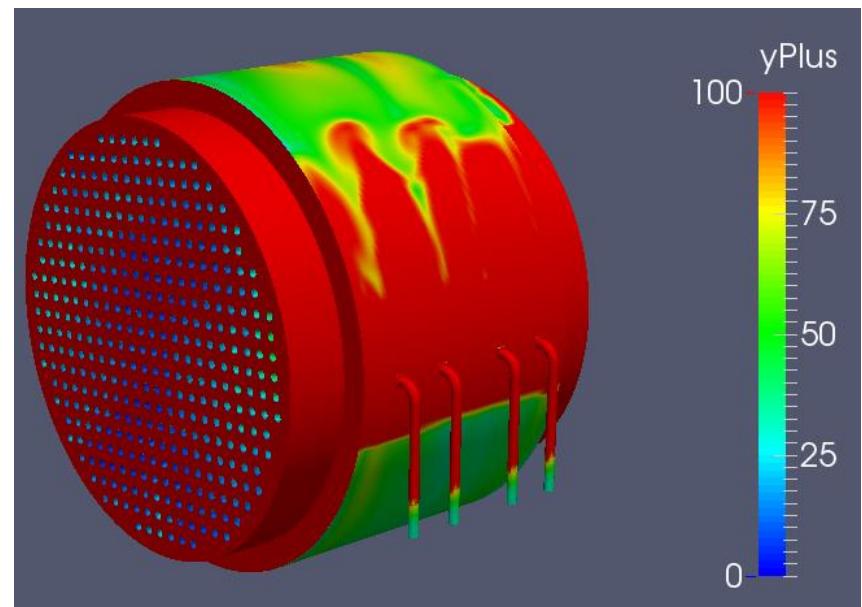
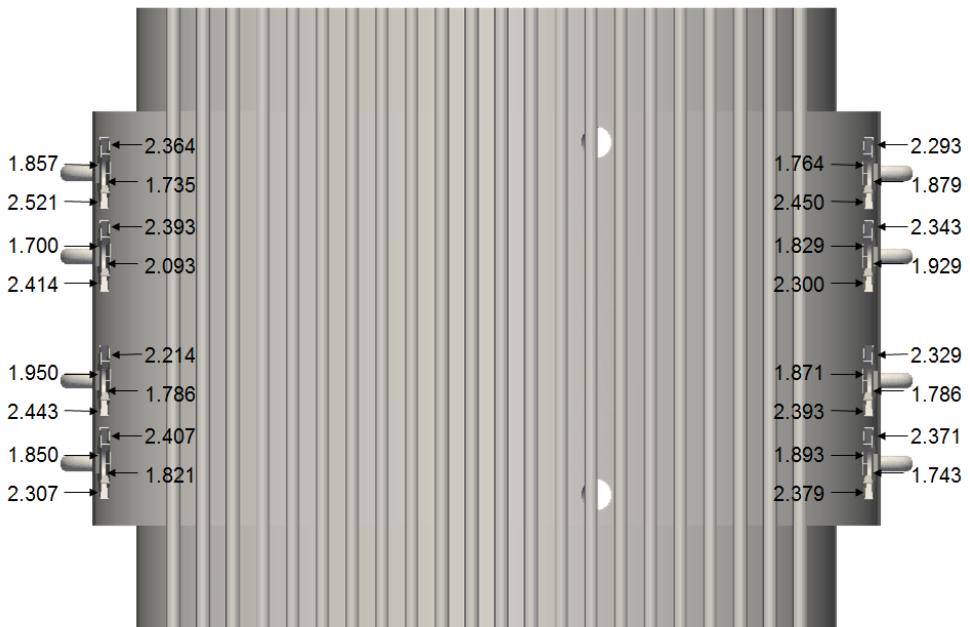
Cross-sectional



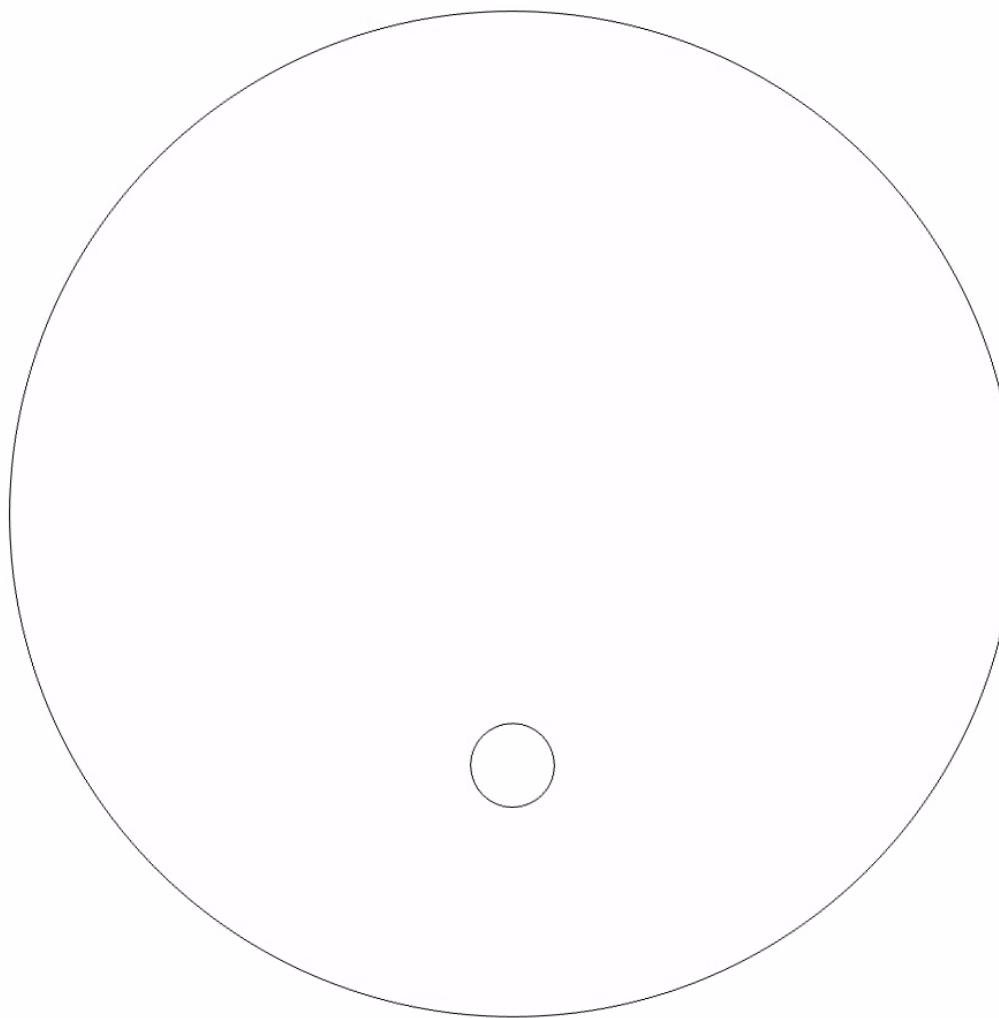
Axial



Velocity at Nozzles & y^+



Animation: Natural Convection



Summary and Conclusion

- 1) 1-D System Code(RELAP/MARS)
→ 3-D CFD Code
(In-House/Commercial/**Open Source**)
- 2) Validation of STERN Lab. Exp.: **Feasibility O.K.**
- 3) Simulation of Full-scale CANDU-6 Moderator:
 - Two-stage computation/pseudo-steady
 - Use of SIMPLE and PIMPLE Foams
 - Realizable $k-\epsilon$ Turbulence Model
 - No Converged Solution: Natural Characterisitcs of N-S Equations
 - Max. temperature: **89 degC**



Q & A

Thank you for your attention!

